# MEASUREMENT OF MEAN VELOCITY PROFILE IN A TURBULENT PIPE FLOW – A NUMERICAL APPROACH

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### ABSTRACT

Computational Fluid Dynamics FLUENT codes were employed in the measurement of the mean velocity profile of a fully-developed pipe flow. The codes used the standard two equation eddy-viscosity ( $k - \varepsilon$ ) model for analysing turbulence in the flow. Deviations of about 2-12%, 1.1-1.7% and 1.02-5.3% obtained between the numerical results and the experimental benchmark for the study in the viscous sub-layer, buffer region and the fully-developed region of the flow show the close agreements between the experimental and the numerical results.

Keywords: Computational Fluid Dynamics, Velocity Profile, Viscosity, Turbulence

**DOI:** <u>https://doi.org/10.54043/laujet.2011.07.01.01</u>

### INTRODUCTION

Well-developed pipe flow profile at measurement positions is required for idealizing the measurement performance of conventional flow meters in the industries. This requirement, which may be in the form of long and straight length of constant-diameter pipe upstream of the measurement points, is important in overcoming the difficulties associated with such measurement.

Whilst the application of the Ultrasonic Velocity Profile (UVP), the Particle Image Velocimetry (PIV) and other measurement techniques has been successful in obtaining the mean velocity profile and Reynolds stress for single-phase laminar, transitional and turbulent flows in pipes and channels (Mori et al, 2002; Yamanaka et al, 2002; van Doorne and Westerweel, 2007; Bai et al, 2010; Odewole et al, 2010), the integrity of these results for application in a wider range of flow measurements are still questionable.

This paper therefore focuses on the use of the standard  $k - \varepsilon$  model available in general-purpose Computational Fluid Dynamics (CFD) FLUENT codes, to estimate the velocity profile in a turbulent pipe flow using the measurement data of Odewole et al, 2010 as an experimental benchmark.

## METHODOLOGY Grid Size and Sensitivity

In order to allow a comparison with the experimental results of Odewole et al, 2010; numerical simulations are carried out on a two dimensional pipe model with a dimension of 102mm x 8976mm, with the dimension of the pipe sufficiently long enough to make the flow fully developed. The flow domain in this study is discretized into 16,800 mesh elements, with a greater mesh density

placed in the vicinity of the pipe walls since the reliability of numerical results for wallbounded flows depends on the grid resolution in the wall regions (Fluent, 2006).

The greater mesh density employed in the near-wall regions of the flow pipe for this study is also particularly important in order to obtain accurate information since the numerical results are susceptible to grid dependency in these regions.

### **Turbulence Model**

The Navier-Stokes equations governing the incompressible, turbulent pipe flow in this study are solved by employing the Reynolds average technique of flow computation since direct numerical simulation places excessive demands on computational resources (time and space) for practical engineering analyses

(Rodgers and Eveloy, 2004). The Reynolds-Averaged Navier-Stokes (RANS) equations shown in Equations (1) and (2) are therefore combined with the standard  $k - \varepsilon$  turbulence model in order to obtain accurate information on the flow. Equation (1) is for the conservation of mass while Equation (2) is the momentum equation in the *x*-, *y*-, and *z*-directions.

$$\frac{\partial}{\partial x_{j}}\left(\rho u_{i} u_{j}\right) = -\frac{\partial P}{\partial x_{i}} + \frac{\partial}{\partial x_{j}}\left[\mu\left(\frac{\partial u_{i}}{\partial x_{j}} + \frac{\partial u_{j}}{\partial x_{i}} - \frac{2}{3}\delta_{ij}\frac{\partial u_{l}}{\partial x_{l}}\right)\right] + \frac{\partial}{\partial x_{j}} - \frac{2}{3}\delta_{ij}\frac{\partial u_{l}}{\partial x_{l}}\right] + \frac{\partial}{\partial x_{j}} - \frac{2}{3}\delta_{ij}\frac{\partial u_{l}}{\partial x_{l}} + \frac{\partial}{\partial x_{j}}\frac{\partial u_{l}}{\partial x_{j}} + \frac{\partial}{\partial x_{j}}\frac{\partial u_{l}}{\partial x_{l}}\right] + \frac{\partial}{\partial x_{j}}\frac{\partial u_{l}}{\partial x_{j}} + \frac{\partial}{\partial x_{j}}\frac{\partial u_{l}}{\partial x_{l}} + \frac{\partial}{\partial x_{l}}\frac{\partial u_{l}}{\partial x_{l}} + \frac{\partial}{\partial x_{l}}\frac{\partial u_{l}}{\partial x_{l}} + \frac{\partial}{\partial u_{l}}\frac{\partial u_{l}}\frac{\partial u_{l}}{\partial x_{l}} + \frac{\partial}{\partial u_{l}}\frac{\partial u_{l}}\partial$$

The need for turbulence modelling, in addition to solving the RANS equations above has grown tremendously over the years. This is as a result of the need to capture the different sizes of turbulent eddies, responsible for the transport and mixing of turbulence in fluid flow, on an appropriate scale. The Reynolds stress,

 $-\rho u_i' u_j'$  in Equation (2) must be properly modelled for closure of Equations (1) and (2). The form of closure obtained depends on the type of turbulence model used.

Using the two equation  $k - \varepsilon$  model,  $-\rho u_i' u'_i$  can be related to the mean velocity gradients within the pipe using the Boussinesq approach as shown in Equation (3) (Versteeg and Malalasekera, 1995).

Viscous dissipation and radiation are not considered here.

$$\frac{\partial}{\partial x_i} (\rho u_i ) = 0, \tag{1}$$

$$\frac{2}{3}\delta_{ij}\frac{\partial u_l}{\partial x_l}\Big|_{j=1}^{j=1}+\frac{\partial}{\partial x_j}\frac{(i')}{(im)}\Big).$$
(2)

$$\frac{1}{1-\rho u' u'} = \mu \left[ \frac{\partial u}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right] - \frac{2}{3} \left[ \rho k + \mu \left[ \frac{\partial u_k}{\partial x_k} \right] \right]$$
(3)

The advantage of the Boussinesq approach lies in the relatively low computational cost associated with the evaluation of the turbulent viscosity,  $\mu_t$  (Rodgers and Eveloy, 2004; Fluent, 2006). With the  $k - \varepsilon$ turbulence model, two additional equations for the turbulent kinetic energy, k and the turbulent dissipation rate, E are solved.

$$\frac{\partial}{\partial x_i}(\rho ku) = \frac{\partial}{\partial x_i} \begin{bmatrix} \mu_{\mu} & \mu_{\mu} \\ \mu_{\mu} & \mu_{\mu} \end{bmatrix} + G_{\mu} + G_{\mu} + G_{\mu} - \rho \varepsilon.$$

$$\frac{\partial}{\partial x_i} \left( \rho \varepsilon u_i \right) = \frac{\partial}{\partial x_j} \left[ \begin{pmatrix} \mu \\ \mu \\ \mu \\ \mu \\ \mu \end{pmatrix} - \frac{\partial}{\sigma_{\varepsilon}} \right] \frac{\partial \varepsilon}{\partial x_j} \left[ + C \right]_{\varepsilon} \frac{\varepsilon}{k} \left( G_k + C G_j \right) - C \sum_{2\varepsilon} \rho \frac{\varepsilon^2}{k} \right] .$$
(5)

The  $k - \varepsilon$  turbulence model, since its introduction by Launder and Spalding (1974), has found extensive use in industries because of their modest requirements for computing resources, robustness, and their performance in obtaining fairly accurate predictions in most flows.

Further readings and information on the  $k - \varepsilon$  model can be obtained from the following literatures: Nallasamy, (1987); Menter, (1994) and Fluent, 2006.

#### **Numerical Method**

The discretized forms of the RANS equations in (1) -

(3) above and Equations (4) and (5) for the turbulence employed in this study were solved on a control-volume basis using a pressure-based solver, where a segregated solution algorithm was used. The momentum and other scalar variables constituting the convection terms on the cell faces of Equations (1) - (5) were obtained using the second order upwind scheme in order to reduce numerical diffusion while the second-order accurate central-differencing scheme was employed for the diffusion terms of the governing equations.

Since FLUENT adopts a co-located scheme, where pressure and velocity values are both stored at centres of computational cells, it therefore implies that cell-centred pressure values also have to be interpolated to obtain pressure values at the face of cells. To do this, the

second order pressure interpolation scheme was adopted. In order to model the interaction between pressure and velocity, pressure correction was carried out on Equation (1) so as to couple together pressure and velocity values in Equations (1) and (2) using the SIMPLE (Semi-Implicit Method for Pressure-Linked Equations) algorithm for pressure-velocity coupling described by Patankar (1980).

Using the size of the flow inlet as a dominant length scale in the description of the flow regime in a nondimensional form, a Reynolds number of 6800 was used as the basis for the selection of the appropriate turbulence model à priori. Consequently, the two-

equation, eddy-viscosity model (  $k - \varepsilon$  ) was then used for modelling turbulence in the pipe flow. In order to bridge the viscosity-affected near-wall region and the fully turbulent core of the pipe together, a two-layer based, non-equilibrium wall function was used. This wall function sensitized the Launder and Spalding's logarithmic law for mean velocity to pressure-gradient effects in the flow (Launder and Spalding, 1972).

The kinetic energy of turbulence in near-wall cells is computed based on the assumption that the flow in the pipe can be regarded as made of a viscosity-affected near-wall region and the fully turbulent core. Thus, the

use of the non-equilibrium wall functions effectively relaxes the local equilibrium assumption adopted by the standard wall functions in computing the budget of kinetic energy of turbulence at near-wall cells (Menter, 1994 and Nallasamy, 1987). Convergence of the steady, incompressible turbulent pipe flow was checked by

ensuring that the scaled residual,  $R^{\phi}$  in Equation (6)

below, is less than  $10^{-4}$  for all variables. User modifications, where applicable, were made to the implementation of the turbulence model employed in the FLUENT code.

$$R^{\phi} = \frac{\sum_{cells,C} \left| \sum_{nb} a_{nb} \phi + b_{o} - a_{C} \phi_{C} \right|}{\sum_{cells,C} \left| a_{C} \phi_{C} \right|} \quad . \tag{6}$$

### **RESULTS AND DISCUSSION**

A comparison between the mean velocity profiles obtained from the numerical measurement carried out in this study and the experimental data of Odewole et al, 2010 is depicted in Figure 1 below. It is evident from Table 1 that a good agreement exists between the experimental and the numerical data in the viscosity-affected inner layer of the pipe since a deviation of about 2-12% is obtained. This viscosity-affected inner layer, where the viscous force predominates the inertia force, lies between y = 0 (the wall of the pipe) and a distance y = 0.41mm away from the wall, where y is measured in wall units.



Figure 1: Comparison of Linear Velocity Profiles

Table 1: Mean Ve	locity Profile	Measurements
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Exp	erimental	Nun	nerical	
Mea	asurement	Meas	urement	Deviation
(Ode	wole et al,			
	2010)	(Prese	nt Study)	
U/Um	1-r/R	U/Um	1-r/R	(%)
0.17	0.0020	0.15	0.0000	11.76471
0.31	0.0039	0.30	0.0040	3.225806
0.49	0.0078	0.48	0.0080	2.040816
0.58	0.0118	0.57	0.0120	1.724138
0.66	0.0196	0.65	0.0200	1.515152
0.70	0.0294	0.69	0.0300	1.428571
0.72	0.0392	0.71	0.0400	1.388889
0.74	0.0490	0.73	0.0500	1.351351
0.76	0.0588	0.75	0.0600	1.315789
0.79	0.0784	0.78	0.0800	1.265823
0.81	0.0980	0.80	0.1000	1.234568
0.83	0.1176	0.82	0.1200	1.204819
0.87	0.1569	0.86	0.1600	1.149425
0.91	0.1961	0.90	0.2000	1.098901
0.94	0.2941	0.92	0.2900	2.12766
0.98	0.3922	0.95	0.3900	3.061224
0.94	0.5882	0.99	0.5900	-5.31915
0.98	0.7843	0.99	0.7800	-1.02041
1.00	1.0000	1.00	1.0000	0

Whilst the CFD codes can predict accurately the velocity variation in the flow within this region, the inability of the hot-wire probe employed in the experimental measurement of the velocity profile in obtaining a good fit of data for the velocity variation within this region lies in the geometry of the hot-wire probe. The pre-calibrated straight hot-wire probe is unable to reach the wall of the pipe, with the measurement of the velocity profile starting at a distance corresponding to y = 10mm from the wall. This causes restricted data for the velocity variation within the viscosity-affected inner layer of the pipe to be obtained. Further information on the experimental procedure and measurement can be obtained from Odewole et al, 2010. However, a higher mesh density positioned at the wall and within the vicinity of the wall ensures that adequate information for the velocity variation within the viscosity-affected inner layer can be obtained from the numerical investigations.

A very good agreement is also obtained between the experimental and numerical data in the buffer region of the flow where the turbulent eddies are rapidly damped and consequently, the turbulent shear stress is reduced to practically zero levels in comparison to the viscous stress. This zone, corresponding to the region between y = 0.61mm to y = 10.2mm in Figure 1, shows a deviation of 1.1-1.7%. Beyond the buffer region, the contribution of viscosity decreases till equal contribution is obtained between the viscous and the inertia forces in the flow. This region, corresponding to y = 10.2mm and y = 10.2

51mm (the centreline of the pipe), is called the fullydeveloped region of the flow. A maximum deviation of about 5.3% exists between the experimental and the numerical data in this region.

#### CONCLUSION

In this paper, the velocity profile of a turbulent pipe flow has been measured numerically using FLUENT while turbulence in the pipe flow was modelled using

the standard  $k - \varepsilon$  model as a result of its robustness in promoting stable convergence and universality for most practical engineering analyses. Deviations of about 2-12%, 1.1-1.7% and 1.02-5.3% obtained in the viscosityaffected inner layer, buffer region and the fullydeveloped region of the pipe flow in comparison with the experimental results are evidence of the close agreements between the numerical results and the experimental benchmark employed for the study. An alternative method of measuring the velocity profile of conventional flow meters is therefore provided by the results of this numerical study. Using this measurement technique, savings in cost and time are obtained without the need for recourse to experimental measurement techniques requiring high capital outlay.

#### AKNOWLEDGEMENT

The authors greatly acknowledge The University of Manchester, UK for permission to use FLUENT for this numerical study.

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		Unit
	NOMENCLATURE General	-
Quantity	Description	-1
$x_i, x_j, x_k, x_l$	Coordinate axes	ms -
U; U; Uk Ul	Mean velocity components	ms-1
$u_l, u_j, u_K, u_l$	Noul volocity components	Nm-2
$u_i'$ , $u'_j$	Fluctuating velocity components	m28-2
Р	Fluid pressure	-
k	Turbulence kinetic energy	
a	Equation coefficient	-
$b_o$	Net source term	kgm <sup>-3</sup> s <sup>-3</sup>
$G_k$	Generation of turbulent kinetic energy due to	kgm <sup>-1</sup> s <sup>-3</sup>
	mean velocity gradients	0
$G_h$	Generation of turbulent kinetic energy due to	

	buoyancy	
$\mathcal{C}_{1\varepsilon}, \mathcal{C}_{2\varepsilon}, \mathcal{C}_{3\varepsilon}$	Empirical constants	-
$R^{\phi}$	Convergence indicator	-
Quantity	Greek Letters Description	Unit
ρ	Fluid density	kgm <sup>-3</sup>
$\mu, \mu_t$	Dynamic and turbulent viscosities	kgm <sup>-1</sup> s <sup>-1</sup>
O ij	Kronecker delta	-
ε	Turbulence dissipation rate	m <sup>2</sup> S <sup>-3</sup>
$\phi$	Transport scalar variable	-
$\sigma_k, \sigma_\varepsilon$	Turbulent Prandtl numbers for k and $\boldsymbol{\mathcal{E}}$	-
	Dimensionless Quantities	
Quantity	Description	Unit $\operatorname{Re}(=\rho u D \mu)$
	Reynolds number	-
	Subscripts	
Quantity	Description	Unit
nb	Neighbour cells	-
С	Central node on control volume -	
cells	Control volumes	-